



Research Paper

CFD MODELLING OF FINNED TUBE GAS COOLER FOR TRANSCRITICAL OPERATION OF CO₂ SYSTEMS

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Refrigerators and heat pumps play an important role in this modern world. Refrigerants are used as working fluid in the thermal systems. These refrigerants undergo phase change in evaporator to provide a cooling effect. Chlorofluorocarbons and Hydrofluorocarbons are the commonly used refrigerants but their stability in the atmosphere and their effect on GWP and ODP is a concern. Because of the harmful effects of synthetic refrigerants on the environment, industries are now looking for technologies based on natural refrigerants. Because of non-toxicity, high availability in nature and some unique thermo-physical property CO₂ is considered as the replacement for synthetic refrigerants. This fact motivated many and has led to the development of trans critical cycle where the condenser gets replaced by a gas cooler. Use of the gas cooler with heat rejection taking place with large temperature glide offers several unique possibilities such as simultaneous cooling and heating, heat pump drying, water heating, etc. Main difference of CO₂ with other refrigerants is its low critical temperature. However this gives an advantage of high volumetric capacity, which makes the CO₂ systems more compact. Initially micro-channel heat exchangers are suggested as gas coolers for CO₂ systems, but its high manufacturing cost makes it economically non-feasible. The objective of this work is to carry out CFD studies on finned tube gas cooler for transcritical operation of CO₂ systems. Flow pattern, Temperature distribution and performance variations with different operating conditions (mass flow rate of CO₂, air velocity, etc.) are studied and validated by test data obtained from the literature.

Keywords: CO₂, Gas cooler

INTRODUCTION

Two international agreements Montreal Protocol and Kyoto Protocol were introduced

to reduce the problem of ozone layer depletion and global warming. The Montreal Protocol was introduced in September 1987 to phase-

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out the use of Ozone Depleting Substances. Ozone Depleting Potential (ODP) is the fraction of the ozone depleting potency of a substance compared to that of R11 or R12. The Kyoto Protocol was introduced at the third conference of United Nations Framework Convention on Climate Change (UNFCCC) in December 1997, which has imposed restrictions on refrigerants on the basis of GWP. Global warming Potential (GWP) is an index that relates the amount of gas trapped by a greenhouse gas to that of CO₂ over a 100-year period. The CFC refrigerants, although once considered to be the best refrigerants, were abandoned due to high ODP. In this situation, industries are looking for long-term solution for refrigerant related problems. This has triggered a large number of studies to develop new technologies. Instead of continuing search for new refrigerants, there is an increasing interest in technologies based on natural refrigerants. Natural Refrigerants are those substances which are already there in large quantity in the biosphere and which we know as harmless. Because of non-toxicity and non-flammability, carbon dioxide offers both environmental and personal safety. Carbon dioxide (CO₂), has been attracting more and more attention in the HVAC industry. Compared with the conventional refrigerants like R22, R134a and R404A, CO₂ is more Eco friendly with zero ozone-depleting potential and very low direct global warming potential. Along with eco-friendliness, CO₂ based systems have various advantages over conventional systems such as, compatibility with normal lubricants, common machine construction materials, non-flammability, non-toxicity, high volumetric refrigerant capacity, reduced

compression ratio, easy availability and excellent heat transfer properties. However, CO₂ has a quite high operating pressure because of its low critical temperature (31.1 °C) and high critical pressure (73.8 bar). In a CO₂ refrigeration system when heat is rejected to ambient air at temperatures or above 31.1 °C (critical temperature of CO₂) the refrigeration cycle is said to operate in a transcritical state. The conventional air-cooled condenser is therefore replaced with a gas cooler. CO₂ cycle offers several unique applications such as simultaneous heating and cooling, high temperature heating and space heating, etc.

Even though there have been considerable prior research done in the area of cycle analysis and application areas, there appears to be some untouched areas like CFD analysis, flow modelling etc. Many research works have done on the possibility of using micro-channel heat exchangers as Gas Cooler for CO₂ systems. However the design and manufacturing of micro channel gas cooler requires huge investment. Because of this, its use is limited to high end applications only. So many researchers are working on the possibility of using finned tube heat exchanger as a gas cooler for transcritical operation of CO₂ systems. This work is concerned with the CFD analysis of the finned tube gas cooler for transcritical operation of CO₂ systems. Finned tube gas cooler with smooth internal surface have been simulated to study its performance.

METHODOLOGY

An identical gas cooler geometry as used by used by Hwang *et al.* (2005) (Figure 1) is selected for CFD modeling. A periodic section

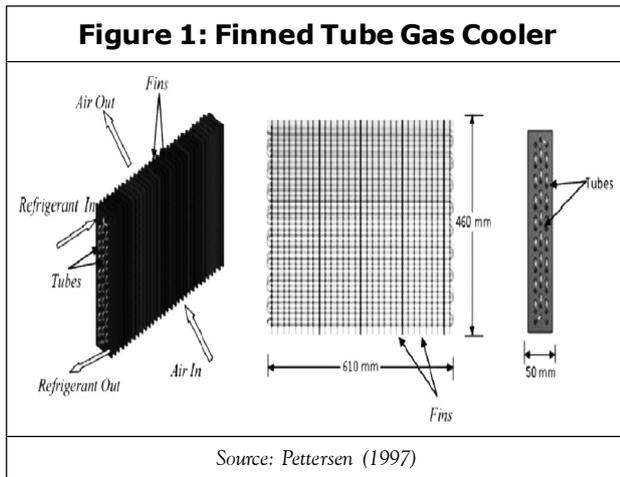
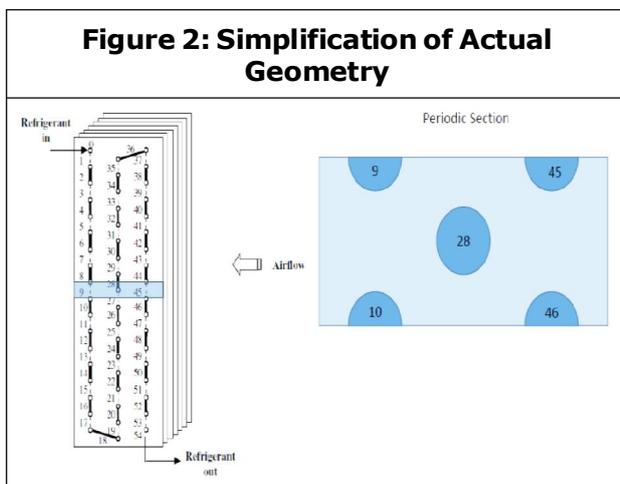


Table 1: Specification of Tested Gas Cooler

| | |
|----------------------------|--|
| Width × Height × Depth | 610 × 460 × 50 mm ³ |
| Front area | 610 × 460 mm ² |
| Fin (plate type) | |
| Material | Aluminum ($K = 250 \text{ W m}^{-1} \text{ K}^{-1}$) |
| Pitch | 1.5 mm |
| Thickness | 0.13 mm |
| Tube (circular and smooth) | |
| Number of row | 3 |
| Number of tube per row | 18 |
| Outside diameter | 7.9 mm |
| Inside diameter | 7.5 mm |
| Material | Copper ($K = 385 \text{ W m}^{-1} \text{ K}^{-1}$) |

five tubes in the middle of the heat exchanger as shown in Figure 2 is and I used to evaluate the performance of finned tube heat exchanger for the trans-critical operation of CO₂. This simplification helped to reduce the overall



count of the mesh. Also it gives the freedom to go for more fine and good quality mesh which will help to get to better results.

CFD DOMAIN AND BOUNDARY CONDITIONS

Domain has been simplified as discussed above and the CAD model for the same is developed using solidworks 2012. To reduce the mesh count only three fins are modelled, two monitor planes are created between first and second fins, between second and third fins and temperature drop between this monitor planes are used as a parameter to judge the performance of the geometry. For the flow to become full developed before entering the CFD domain CO₂ is allowed to flow through a tube without fins. This tube walls are modelled as adiabatic walls so that only turbulence and flow equation will come to the picture for the flow of CO₂ outside the actual CFD domain. Fins and tubes are modeled as surfaces and thickness is given by using shell conduction option in fluent.

An enclosure is made which is covering this arrangement and sufficient length is provided to avoid the possibility of reverse flow. Considering the small gap between each fins normal tetrahedron mesh may not be sufficient for effectively capturing the flow. So wedge (prism mesh with triangular base) mesh is created for the geometry using ICEM CFD 14. Variable height hexahedral mesh is used near the tube surface for effectively capturing the boundary layer formation. First cell height for the same is calculated based on y+ criteria. Once the meshing is done mesh file is imported to fluent 14 boundary conditions are applied to the parts and it is solved for Flow,

Table 2: Boundary Conditions

| S. No. | Parts | Type of Boundary Condition |
|--------|------------------------|----------------------------|
| 1. | Air Inlet | Velocity Inlet |
| 2. | Air Outlet | Pressure outlet |
| 3. | CO ₂ Inlet | Mass flow inlet |
| 4. | CO ₂ Outlet | Pressure outlet |
| 5. | Enclosure Sides | Symmetry |
| 6. | Monitor Surfaces | Interior |
| 7. | Fins, Tubes | Wall |

solution is converged it is solved for turbulence and then for energy equation. Solution is said to be converged when then the energy residuals are in the range of 10⁻¹⁰ and other residuals in the range of 10⁻⁵. Also flatness CO₂ temperature monitors, y+ value, mass and energy balance are also considered to check the correctness of a solution obtained.

Table 3: CFD Result (V_a = 1 m/s, T_{atm} = 29.4 °C)

| tube no | monitor_in | monitor_out | diff | predicted temp drop for full tube(degre cel) | temp drop predicted for the full tube by Y.T. Ge (degre cel) [1] | actual temp drop for the full tube(degre cel) [19] | percentage error_CFD(%) | percentage error_Y.T. Ge(%) | Average Temp drop per fin |
|---------|------------|-------------|----------|--|--|--|-------------------------|-----------------------------|---------------------------|
| 9 | 60.9684 | 60.95188 | 0.016514 | 3.401884 | 2.53 | 3.29 | -3.400729483 | 23.10030395 | 0.007186 |
| 10 | 57.97616 | 57.96371 | 0.012451 | 2.564906 | 2.53 | 2.53 | -1.379683794 | 0 | |
| 28 | 43.00233 | 43.0034 | -0.00107 | -0.220832 | 0.1428 | -0.214 | -3.192523365 | 166.728972 | |
| 45 | 41.49896 | 41.4985 | 0.000458 | 0.094348 | 0.133 | 0.133 | 29.06165414 | 0 | |
| 46 | 41.29908 | 41.29868 | 0.000393 | 0.080958 | 0.133 | 0.133 | 39.12932331 | 0 | |

Turbulence and Energy Equations. For the first simulation identical conditions as given by Ge and Cropper (2009) is used and this results are used to validate the model.

An open source data base called COOL-PROP is used to find the property values of CO₂ at different temperatures. A simple code is developed in MATLAB for this purpose. After defining boundary conditions monitors are set to monitor convergence of temperature CO₂ at monitor planes. From literatures it is found that *k-ε* model is will give better turbulence modeling for fin and tube heat exchanger. So for better accuracy *k-ε* realistic algorithm is used to model turbulence flow of CO₂ in tubes.

Calculations are run till the solution get converged. First only the flow equation is considered for the calculation. Once the

Figure 3: Temperature Contour of Air When Inlet Air Velocity is 1 m/s, 2 m/s and 3 m/s Respectively(T_{atm} = 29.4 °C)

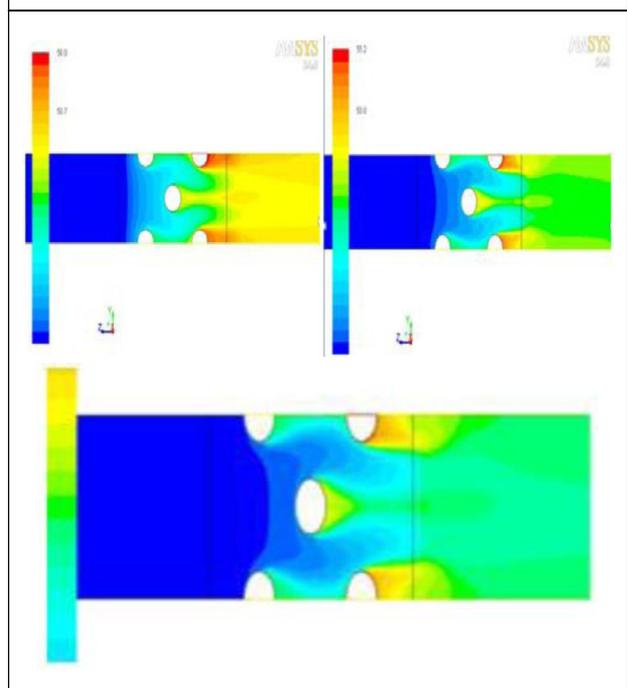


Figure 4: Velocity Contour of Air When Inlet Air Velocity is 1 m/s, 2 m/s and 3 m/s Respectively ($T_{atm} = 29.4\text{ }^{\circ}\text{C}$)

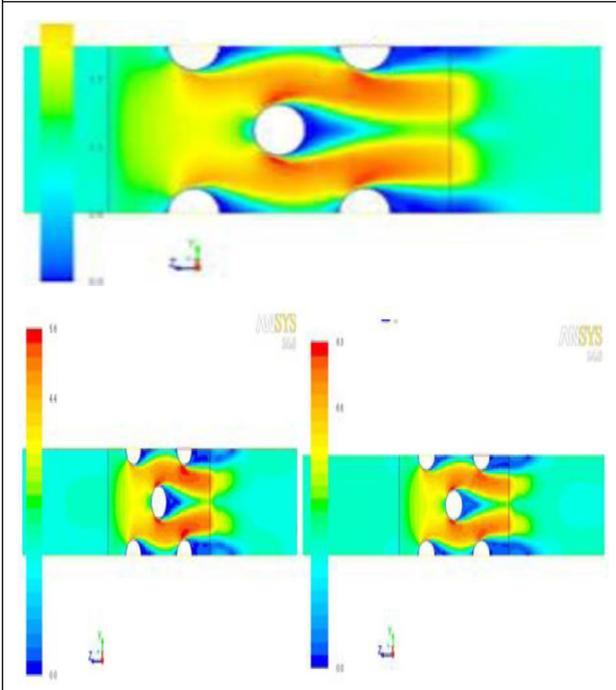
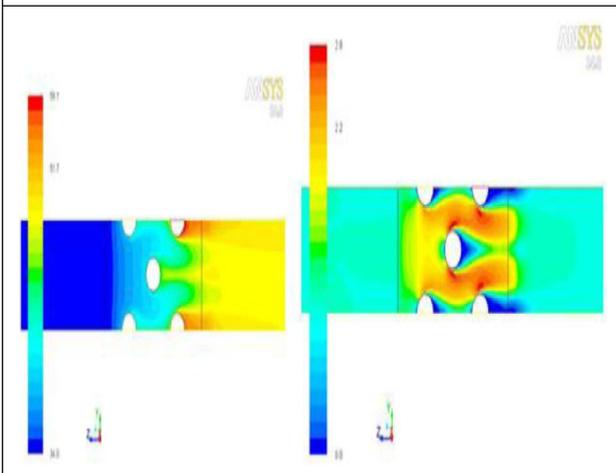


Figure 5: Temp and Velocity Contour of Air When T_{atm} is $34\text{ }^{\circ}\text{C}$ (Inlet Air Velocity = 1 m/s)



RESULTS AND DISCUSSION

The obtained result compared with previously published experimental and simulation result. The CFD results obtained are found to have close matching with experimental result.

The contours displayed in Figures 3 to 5 reveal the temperature increases in the fluid due to heat transfer from the tubes. The hotter fluid is confined to the near-wall and wake regions, while a narrow stream of cooler fluid is convected through the tube bank. The consequences of different air velocity to the fluid temperature distribution are shown in the above figures. When the fluid through the bank of tubes the maximum velocity occur at either the transverse plane or the diagonal plane where the flow area is minimum. The velocity vectors displayed in Figure 3 reveal that the CFD results show very low velocity values adjacent to the tube surface. In the regimes between the tubes, i.e., at the transverse plane the maximum velocity was occurred.

CONCLUSION

CFD gave better results than the Mathematical simulation done by Ge and Cooper(2009) . Effect of eddies formation and flow separation are found to be increasing with air velocity. Flow separation was maximum when air velocity in 2 m/s. There was not much change in this behavior with the change in atmospheric temperature. Temperature of the CO_2 which is flowing through the middle tube is found to be increasing slightly. This is because of the high temperature of CO_2 flowing through nearby tubes. This problem can be overcome by providing more pitch or more air velocity. Temperature profile and velocity profile air was more smoother at higher velocities of air. ☺

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