ISSN 2278 – 0149 www.ijmerr.com Vol. 1, No. 2, July 2012 © 2012 IJMERR. All Rights Reserved

Research Paper

SOUND SPECTRUM MEASUREMENTS IN DUCTED AXIAL FAN UNDER STALL CONDITIONS AT FREQUENCY RANGE FROM 9000 HZ TO 9600 HZ

Manikandapirapu P K1*, Srinivasa G R2, Sudhakar K G3 and Madhu D4

*Corresponding Author: Manikandapirapu P K, 🖂 manikandapirapu83@gmail.com

Performance of axial fan is found to reduce drastically when instability is encountered during its operation. Performance of an axial fan is severely impaired by many factors mostly related to system instabilities due to rotating stall and surge phenomenon experienced during its operation. The present work involves measuring the sound spectrum measurements in ducted axial fan under stall conditions at frequency range from 9000 Hz to 9600 Hz. Objective of the experiment is to measure the frequency domain signal and study the sound Characteristics in ducted axial fan by using spectrum analyser. Different types of FFT signals have been measured under stall conditions for the frequency range of 9000 Hz to 9600 Hz with respect to rotor speed and different graphs are plotted for ducted axial fan.

Keywords: Microphone, BNC connector, Data Acquisition System, LABVIEW, Spectrum Measurements, Throttle position, Rotor speed

INTRODUCTION

Mining fans and cooling tower fans normally employ axial blades and or required to work under adverse environmental conditions. They have to operate in a narrow band of speed and throttle positions in order to give best performance in terms of pressure rise, high efficiency and also stable condition. Since the range in which the fan has to operate under stable condition is very narrow, clear knowledge has to be obtained about the whole range of operating conditions if the fan has to be operated using active adaptive control devices. The performance of axial fan can be graphically represented as shown in Figure 1.

¹ Mechanical Department, Dayananda Sagar College of Engineering, Bangalore.

² Dayananda Sagar College of Engineering, Bangalore.

³ CDGI, Indore, Madhya Pradesh .

⁴ Mechanical Department, Government Engg. College, KRPET-571426.



TEST FACILITY AND INSTRU-MENTATION

Experimental setup, fabricated to create stall conditions and to introduce unstall conditions in an industrial ducted axial fan is shown in Figures 2 to 5.

A 2 HP Variable frequency 3-phase induction electrical drive is coupled to the electrical motor to derive variable speed ranges. Schematic representation of ducted fan setup is shown in Figure 6.





Figure 4: Variable Frequency Drive for Speed Control



Figure 5: Automatic Throttle Controller





The flow enters the test duct through a bell mouth entry of cubic profile. The bell mouth performs two functions: it provides a smooth undisturbed flow into the duct and also serves the purpose of metering the flow rate.

The bell mouth is made of fiber reinforced polyester with a smooth internal finish. The motor is positioned inside a 381 mm diameter x 457 mm length of fan casing. The aspect (L/D) ratio of the casing is 1.2. The hub with blades, set at the required angle is mounted on the extended shaft of the electric motor. The fan hub is made of two identical halves. The surface of the hub is made spherical so that the blade root portion with the same contour could be seated perfectly on this, thus avoiding any gap between these two mating parts. An outlet duct identical in every way with that at inlet is used at the down stream of the fan. A flow throttle is placed at the exit, having sufficient movement to present an exit area greater then that of the duct.

BASIC SOUND SPECTRUM ANALYSER SYSTEM

Basic sound Spectrum analyzer schematic diagram consists of various components as shown in Figure 7. Microphone acquires the sound pressure fluctuation and converts them to an analog signal. BNC connector sends the signal to Data acquisition system. Data Acquisition system receive the signal from the BNC connector and sends to LABVIEW software. Once the amplitude of the signal has been measured, the computer system displays the measurement signal of spectrum through LABVIEW software.



SOUND SPECTRUM ANALYSER

Experimental setup of Spectrum analyser consists of various components is shown in Figure 8. Microphone acquires the sound signals, frequency range from 0 Hertz to 10000 Hz and measure the decibel range from 0 to 130 decibel. Microphone sensitivity is the ratio of its electrical output to the sound pressure at the diaphragm of the microphone. Since a microphone output is usually measured in mill volts (mv) and sound pressure is measured in Pascals. The unit of sensitivity of microphone is mv/Pa. microphone connects to BNC connector. BNC connector transmits the signal to DAQ system. DAQ card consists 2 channel input port to acquire the signal and send the signal to system achieve through LABVIEW

software inbuilt with National Instruments noise and vibration acquisition system and 2 channel output port to receive the signal from the system and to make a active feedback control



SOUND SPECTRUM MEASURE-MENTS

Experiments were carried out to examine the nature of sound pressure variations in a ducted axial fan under stall condition for the frequency range from 9600 Hz to 9600 Hz at throttle position of 3 cm and rotor speed varying from 2400 rpm to 3600 rpm by using spectrum analyser. The variation in sound pressure amplitude level as a function of rotor speed at throttle position 3 cm from the casing is shown in Figure 9. In an axial fan setup, eight number of axial fan blades have been transferred the energy to the fluid. In one rotation of axial fan rotor, aerofoil cross section of axial fan blades transfers the energy to the fluid achieving through by lift force. Lift force of Newton across the fan blade area develop the pressure rise to the fluid as per the operating condition and design condition of axial fan. In a single rotation of blade of ducted axial fan, eight times sound

pressure amplitude value will raise and down periodically. For a same rotation, increase the sound pressure amplitude through energy transfer of blade to fluid. Two blades between the gap fluids enter freely, for that no energy transfer between fluids to blade. So, decrease the sound pressure amplitude in between the gap of blades. Every rotation, rotor speed varies from 2400 to 3600 rpm and throttle position of 3 cm from the casing, with respect to blade passing frequency sound pressure amplitude will rise suddenly up and abruptly falls down in ducted axial fan periodically. A variation in sound pressure amplitude of air at throttling positions of 3 cm when the rotor rotates at 2400 rpm is shown in fig.9. Maximum sound pressure amplitude is found to be 60 decibels at stall condition and the Minimum sound pressure amplitude is found to be 43 decibels at stall conditions which is attributable to combinatorial effects of blockage in mass flow, rotating stall, periodic vibration due to air flow and excitation of fan blade.

A variation in sound pressure amplitude of air at throttling positions of 3 cm when the rotor rotates at 2700 rpm is shown in Figure 10. Maximum sound pressure amplitude is found to be 62 decibels at stall condition and the Minimum sound pressure amplitude is found to be 52 decibels at stall conditions.

A variation in sound pressure amplitude of air at throttling positions of 3 cm when the rotor rotates at 3000 rpm is shown in Figure 11. Maximum sound pressure amplitude is found to be 64 decibels at stall condition and the Minimum sound pressure amplitude is found to be 36 decibels at stall conditions.







A variation in sound pressure amplitude of air at throttling positions of 3 cm when the rotor rotates at 3300 rpm is shown in Figure 12. Maximum sound pressure amplitude is found to be 72 decibels at stall conditions and the Minimum sound pressure amplitude is found to be 44 decibels at stalled conditions. A variation in sound pressure amplitude of air at throttling positions of 3 cm when the rotor rotates at 3600 rpm is shown in Figure 13. Maximum sound pressure amplitude is found to be 70 decibels at stall condition and the





Minimum sound pressure amplitude is found to be 48 decibels at stall conditions.

CONCLUSION

In this paper, an attempt has been made to measure the sound spectrum in frequency domain for the frequency range from 9000 Hz to 9600 Hz under stall condition with respect to rotor speeds in ducted axial fan by using spectrum analyzer. It is useful to examine the characteristics of stall in ducted axial fan. Further, this work can be extended by working on the mathematical model of sound spectrum study in ducted axial fan. The results so far discussed, indicate that sound spectrum measurements of ducted axial fan is very promising.

ACKNOWLEDGMENT

The authors gratefully thank AICTE (rps) Grant. for the financial support of present work.

NOMENCLATURE

- v_w = Whirl velocity in m/s
- ψ = Pressure ratio
- N = Tip speed of the blades in rpm
- Δp = Pressure rise across the fan in N/m²
- d = Diameter of the blade in m
- ρ_{air} = Density of air in kg/m³
- Lp = Sound Pressure Level in db
- BPF = Blade passing frequency in Hz
- L_N = Normalized Sound Level in db

REFERENCES

 Bram de Jager (1993), "Rotating Stall and Surge Control: A Survey", IEEE Proceedings of 34th Conference on Decision and Control.

- Chang Sik Kang (2005), "Unsteady Pressure Measurements around Rotor of an Axial Flow Fan Under Stable and Unstable Operating Conditions", *JSME International Journal*, Series B, Vol. 48, No. 1, pp. 56-64.
- Day I J (1993), "Active Suppression of Rotating Stall and Surge in Axial Compressors", ASME Journal of Turbo Machinery, Vol. 115, pp. 40-47.
- Dixon S L (1998), Fluid Mechanics and Thermodynamics of Turbo Machinery, 5th Edition, Pergamon, Oxford.
- Epstein A H (1989), "Active Suppression of Aerodynamic Instabilities in Turbo Machines", *Journal of Propulsion*, Vol. 5, No. 2, pp. 204-211.
- Patrick B Lawlees (1999), "Active Control of Rotating Stall in a Low Speed Centrifugal Compressors", *Journal of Propulsion and Power*, Vol. 15, No. 1, pp. 38-44,
- Poensgen CA (1996), "Rotating Stall in a Single-Stage Axial Compressor", *Journal of Turbo Machinery*, Vol. 118, pp. 189-196.
- Paduano J D (1996), "Modeling for Control of Rotating stall in High Speed Multistage Axial Compressor", *ASME Journal of Turbo Machinery*, Vol. 118, pp. 1-10.
- Ramamurthy S (1975), "Design, Testing and Analysis of Axial Flow Fan", M E Thesis, Mechanical Engineering Dept., Indian Institute of Science.
- William W Peng (2008), Fundamentals of Turbo Machinery, John Wiley & Sons. Inc.